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Issues

MANDATORY BULLETIN No. EV 97 – 011 a SPORTSTAR - 011 a

All EV-97 EUROSTAR airplanes, which already reached 2000 1. **CONCERNING TO:**

total hours flown,

as well as SportStar airplanes, which already reached 3000 total

hours flown.

Introduction of a new periodical inspection **REASON:**

after 2000 hours (EV-97 EUROSTAR) and

after 3000 hours (SportStar).

3. REQUIRED ACTION: Perform inspection according to the Appendix 1 of this bulletin

within the scope of the nearest 100 hours inspection.

Record inspection into the Airplane Log Book.

Inform the airplane manufacturer on performance and results of

the inspection.

LATEST DATE OF THE ACTION: within the scope of the nearest 100 hours inspection. 4.

ACTION CARRIED OUT BY: 5. Appropriately authorized person

(depends on the regulations valid in particular country).

COSTS COVERED BY: 6. Airplane owner. 7. **NECESSARY MATERIAL:** see Appendix 1 8. **WORK PROCEDURE:** see Appendix 1

9. **ENCLOSURES:** Appendix 1 – Inspection of EV-97 EUROSTAR Airplane after

2000 hours and SportStar Airplane after 3000 hours.

10. ELABORATED BY: Jan Fridrich Sr., Technical Department

Approved by: Petr Javorsky, LSA Project Manager Date: 9.6.2009

QS-406/F-03A EV97_011a_SP_011a_Inspection_2000_3000h

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Appendix 1 Inspection of EV-97 EUROSTAR Airplane after 2000 hours and SportStar Airplane after 3000 hours.

Туре:	S/N:	Time Since New [hrs]:	Check performed on:		
	Registration mark:	Time Since New [cycles]:			

Action No.	Cubicat of the about	Check ☑ tick off the performed check			
	Subject of the check (Checks for cracks, deformations, wearing,	After 2,000 FH EV-97	After 3000 FH	Carried out by	Checked by
	missing parts, etc.)	Repetition after 500 FH	Repetition after 1000 FH		
	Before start of the check – clean or wash polluted surface of the aircraft, if necessary Check aids: mirror, electric torch, magnifying glass 1:4 or 1:10	×	×		
1.	Fuselage				
1.1.1	Fuselage central section hinges				
1.1.2	Fuselage central section hinges				
1.1.3	Area of connection of caps and webs	×	×		
1.1.4	Area of connection of caps and webs	×	×		
1.1.5	Wing hinges, flap control hinges	×	×		
1.2.1	Stabilizer hinges	×	×		
1.3.3	Main landing gear attachment under seats	×	_		
1.3.4	Main landing gear attachment under seats		×		
1.4.1	Nose landing gear attachment		×		
1.4.2	Nose landing gear attachment	×			
1.5.1	Left upper engine hinge on the fire wall	×	×		
1.5.2	Left bottom engine hinge	×	×		
1.5.3	Attachment of the left upper engine hinge on the fire wall	×	×		
1.6.1	Area of the fuselage reinforcement	×	×		
1.6.2	Area of the fuselage reinforcement	×	×		
1.6.3	Area of the fuselage reinforcement	×	×		
1.6.4	Bottom part of the fuselage under the left bottom engine hinge	×	×		
1.6.5	Check of the rivet joints of the main undercarriage to the wing centre section spar caps	×	×		
1.7.1	Towing equipment hinge	×	×		
2.	Wing		1		
2.1	Bottom part of the wing in the area of wing- fuselage hinges	×	×		
2.2	Access to check of caps through the rib No. 1				
2.3	Wing hinges between ribs 1 and 2	×	×		
2.4	Bottom part of the wing in the area of wing- fuselage hinges	×	×		
2.5	Wing spar with hinges	×	×		
2.6	Wing spar with hinges	×	×		

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		Check ☑ tick off the performed check			
Action No.	Cubiast of the about				
	Subject of the check (Checks for cracks, deformations, wearing, missing parts, etc.)	After 2,000 FH EV-97 Repetition after 500 FH		Carried out by	Checked by
2.7	Upper hinge	1	1		
2.8	Bottom hinge				
2.9	Area of bottom caps with wing-fuselage hinges	×	×		
2.10	Area of upper caps with wing-fuselage hinges	×	×		
3.	Tail unit				
3.1.1	Stabilizer hinges	×	×		
3.1.2	Stabilizer hinges				
3.2	Stabilizer spar	×	×		
3.4	Rudder step bearing	×	×		
3.4.2	Fin spar with rudder hinges	×	×		
4.	Control system	According to the simultaneous 100 hrs check			
5.	Landing gear				
5.1.1	Main landing gear leg		×		
5.1.2	Main landing gear leg critical place	×			
5.1.3	Attachment of the wheel axis with the nose landing gear leg	×	×		
5.2.1	Nose landing gear leg	×			
5.2.2	Nose landing gear leg		×		
6.	Engine installation				
6.2	Engine bed in the area of hinges	×	×		
7.	Electrical instruments	According to the simultaneous 100 hrs check			
8.	Equipment	According to the simultaneous 100 hrs check			

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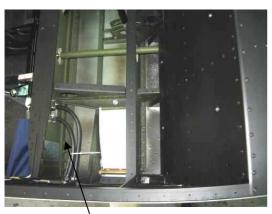
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1.1.1



Fuselage central section hinges, see figures 1.1.3 and 1.1.4

1.1.3



Area of connection of caps and webs (after drilling out of seats) 1.1.5

1.1.2



Fuselage central section hinges, see figures 1.1.3 and 1.1.4

1.1.4



Area of connection of caps and webs (after drilling out of seats)



✓Max. radial play - 0.2 mm

Wing hinges, flap control hinges

Maximum diameters of pins EV-97 Ø16+0.03 mm

SportStar Ø15+0.03 mm

Minimum diameters of pins EV-97 Ø16+0.03 mm

SportStar Ø16-0.03 mm

Maximum diameter of rear hinge Ø8+0.04 mm

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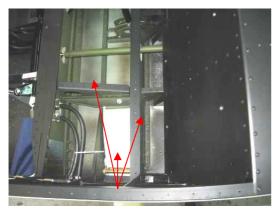
1.2.1 Rudder root hinge



Stabilizer hinges Pin diameters: min Ø6-0.04 mm Ø8-0.05 mm

Check of hinges rivet connections

1.3.4



Main landing gear attachment under seats Check of rivet connections in the area of the main landing gear attachment

1.3.3



Check of rivet connections in the area of the main landing gear attachment

1.4.1



Nose landing gear attachment - SportStar aircraft



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1.4.2



Nose landing gear attachment EV-97

1.5.2



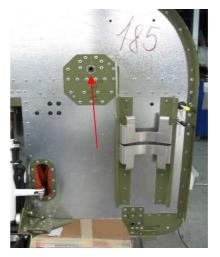
Left bottom engine hinge Check of the hinge and riveted joints (front view)

1.6.1



Area of the fuselage reinforcement Check of the hinge and riveted joints

1.5.1



Left upper engine hinge on the fire wall Check of rivet joints (front view)

1.5.3



Attachment of the left upper engine hinge on the fire wall
Check of the hinge and riveted joints
(view from inside of the fuselage)

1.6.2



Area of the fuselage reinforcement Check of the hinge and riveted joints

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1.6.3



Area of the fuselage reinforcement Check of the hinge and riveted joints

1.6.5



Wing central section joints

1.6.4



Bottom part of the fuselage under the left bottom engine hinge Check of the hinge and riveted joints

1.7.1



Towing equipment hinge Check of condition of rivet joints

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2.1 Bottom part of the wing in the area of wingfuselage hinges



View on the bottom part of the wing in the area Of wing-fuselage hinges

2.3 Wing hinges between ribs 1 and 2



View on wing hinges between ribs 1 and 2 (without front skin)

2.2 Access to check of caps through the rib No. 1



Access to check of caps through the rib No. 1 (only for information)

2.4 Bottom part of the wing in the area of wingfuselage hinges



Rear hinge - max. Ø8+0.04 mm View on the bottom part of the wing in the area of wing-fuselage hinges (unfinished wing, only for info)



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2.5 Critical area of the cap at the end of the hinge



Wing spar with hinges Check of joints

2.7



Upper hinge Area of the wing spar (view against the flight direction, for information only) only)

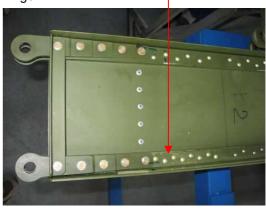
2.9



Area of bottom caps with wing-fuselage hinges

(view from inside through openings in the rib No. 1)

2.6 Critical area of the cap at the end of the hinge



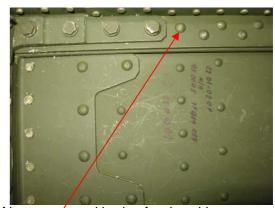
Wing spar with hinges Check of joints

2.8



Bottom hinge Area of the wing spar (view against the flight direction, for information

2.10



Area of bottom caps with wing-fuselage hinges - the last bolt – check of joint (view from inside through openings in the rib No. 1)



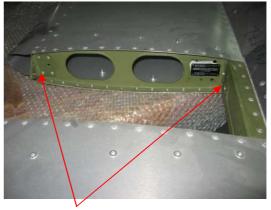
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3.1.1



Stabilizer hinges Check of hinge rivet joints Hinges openings diameters: Ø8+0.04 mm; Ø6+0.03 mm

3.1.2



Stabilizer hinges (for information only)

3.2



Stabilizer spar Check of caps in the central part of the stabilizer

3.4



Rudder step bearing
Rudder step bearing
Check of rivet joints and play
in the rudder hinge (max. 0.05 mm)



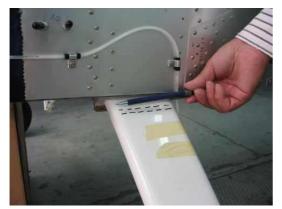
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3.4.2



Fin spar with rudder hinges Check of spar caps (after dismantling of the rudder)

5.1.1



Main landing gear leg - SportStar Check area - under the pencil

5.1.3



Attachment of the wheel axis with the nose landing gear leg

5.1.2



Main landing gear leg - EV-97

5.2.1



Nose fanding gear leg - check of the central guiding area - EV-97

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5.2.2



Nose landing gear leg
- check of the central guiding area - SportStar

6.2



Nose landing gear leg area of hinges on the fire wall and at the engine (the critical place is indicated)